

FINAL REPORT
ON
BALLAST STUDY

Contracts Nonr 2355 (00) and
Project 147
5 December 1957
FOR
OFFICE OF NAVAL RESEARCH
CODE 461
WASHINGTON, D.C.
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FINAL REPORT

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PROJECT 147 - BALLAST STUDY (Including Analysis of Proj. 126 Flights)

I. INTRODUCTION.

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In April of 1957 a project was begun with the object of testing the accuracy of a combined timer-ballast unit which was designed for use on small balloon carriers. The work was done on Raven Industries Project 126 under Contract Nonr 2355(00). The units to be tested were supplied as GFE. Two progress reports have been written covering work done in the periods 1 April to 26 September and 27 September to 6 November 1957.

Thirty flights were to have been made on Project 126. It became clear from the outset that the project would involve more than the mere accumulation of statistical data. In addition to the discovery of a few minor "bugs" in the mechanism, it was found that the required ballast rate was dependent on a number of variables including balloon size, free lift, and atmospheric conditions. Actually, a total of thirty-five flights were made, the majority of which were offorts to pinpoint these variables and establish an optimum ballast rate.

Project funds were nearly exhausted before the results of the exploratory tests could be combined and checked in a final series of flights. For this reason, a new contract was written (dated 1 November 1957) and project 147 was instituted for the purpose of launching six additional balloons whose ballast units were programmed in accordance with the indications of the project 126 flights.

Because of the similarity in purpose of the two projects, the results of both have been tabulated and analyzed for presentation in this final report.

II. TEST PROGRAM.

The program included tests of two sizes of balloons. One, of 130 cu. ft. volume, carried a 2 lb. useful load and floated at approximately 20,000 ft. A 250 cu. ft. cell was used to carry a 4 lb. useful load in the cicinity of 25,000 ft. altitude. Because of a greater need, emphasis was placed on the accumulation of data for the smaller balloon. Twenty-eight such flights were made, as opposed to thirteen utilizing R-250 balloons. The P-130's were all flown in daylight hours, some under clear conditions and some in varying degrees of cloudiness. The R-250's were all flown in clear weather; 8 during the day and 5 at night.

Of the 41 flights which were made, 36 were recovered. This recovery ratio (88%) is remarkable, and may be attributed in large part to

two factors. First, by careful selection of launching conditions, balloons can be sent into favorable recovery areas (flights of 12 September and 12 November are exceptions and were made in spite of poor conditions), and second, the flights were rigged so that the equipment would descend with the balloon attached. This made for greater visibility and eye catching power. The rigging neces to accomplish this consisted of a release hook which disconnected the timer from the base of the balloon (Figure 1) and a separate line running from the timer to a parachute which in turn was suspended from the top of the balloon (Figure 2). The release hook was opened by the jerk resulting when the payload falls. felt that this rigging, which would not be used in an operational situation, would not jeopardize the actual test being conducted. This assumption is believed to have been substantiated, with the exception of one instance (Flt. 189) where the hook opened accidentally early in the flight. Other malfunctions did occur, but none which affected the experimental phase of the flight. In a number of cases the timer failed to release from the balloon (parachute line probably fouled the release hook) and the system descended to the ground still inflated. While this has a much greater eye catching ability, it is undesirable because the additional distance traveled may take the balloon beyond a good recover area. Also, the hydrogen lifting gas presents a hazard to the finder. Late in the project an improved release was developed which was more satisfactory (See Figure 8).

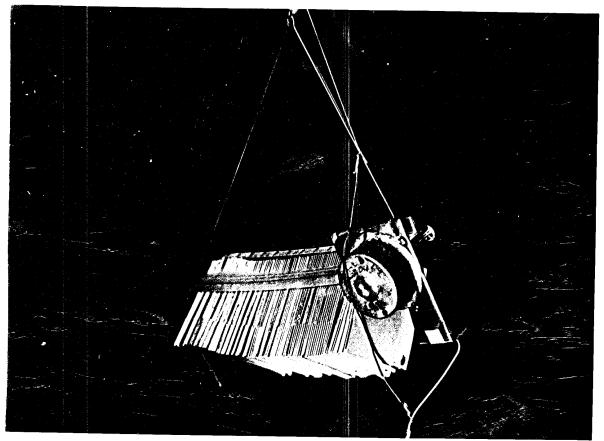


FIGURE 1

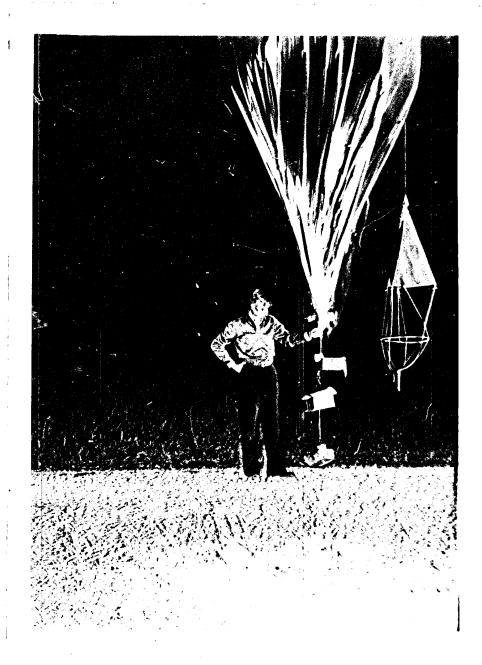


FIGURE 2

At the start of Project 126, flights were made for simple observation of actual duration vs. predicted duration and of altitude maintenance when dropping ballast at a standard rate of 38 grams per hour. It quickly became apparent that although the timer was operating very satisfactorily, the ballast rate would require some modification.

III. TIMER FUNCTION.

As previously mentioned, the timer was found to be very accurate. The error noted has been less than 3% in all except three of the cases where definite termination has been determinable. Of these three, one was 6% off and the others were in error by increments of whole hours. In all probability, the latter two operated properly,

but the error was in setting the timer or recording the planned duration in the flight log.

Of the instances when definite termination was not observable by reason of some malfunction or non-recovery, indications exist in most cases to show that the timer could not have been in error by any large amount. The breakdown on timing accuracy is as follows:

Total flown	41
Accurate within 3%	18
Indications of good	
accuracy	13
Unknown	10

Some of the unknowns are the result of failure in the ballast function or the barograph, while others arise from malfunctions in the timer mechanism itself. Three types of malfunctions were observed and at least two can be eliminated.

These two malfunctions were caused by the slack portion of the load line which is wrapped around the hub of the timer shaft. On Flight No. 198 and possibly No. 178 this loop of line worked into the clock case and fouled the gear train. On several other occasions the loop was left taped in place, both intentionally and accidentally. It was found that this offered sufficient resistance so that the timer would stop about 15-25 minutes before termination was scheduled. This occured on Flights 186, 187, 190 and possibly 184.

On Flight No.188 the timer stopped during ascent. This did not appear to be similar to Flight 198, and no explanation could be found for the stoppage.

In one additional case, the payload struck a telephone pole at launch, spilling the droppable material. The resulting weight on the load line was insufficient to drive the timer. This cannot be considered a timer malfunction.

IV. BALLAST FUNCTION.

Early experience indicated that if a constant rate of ballast release was relied upon, a balloon launched with a moderate free lift would start to descend immediately after reaching peak altitude and generally would not recover itself until it had reached a point 3000-10,000 ft. below the peak. This was noted in flights where P-130's and R-250's had free lifts of 200 Gm. and 250 Gm. respectively. Experience on a previous project also pointed out that the post-ascent dip was to be expected on small balloons of the type used here. The explanation for this probably lies in the thermodynamic effects (or lack of same) on the balloon. As the balloon rises, its lifting gas tends to remain colder than the surrounding atmosphere because of expansion. When a vented balloon reaches ceiling it will valve

its free lift, and in all probability overvalve, halting or reversing its travel. After the ascent has stopped, the gas begins to warm to the ambient temperature, and if it is a daytime flight, solar radiation will warm the gas still further. In large balloons these warming effects are sufficient to counter the effect of overvalving and the balloon will come into equilibrium and will remain at ceiling. This inherent stability appears to be proportional to balloon size and for P-130's and R-250's is insignificant.

Two approaches were made toward the elimination of the post-ascent dip. First, by lowering the free lift, overvalving can be reduced to the point where a ballast rate (38 Gm/hr) will provide an adequate restoring force. In this case a free lift of 100 grams was tried with satisfactory results (Flights 172, 174 and 182). A disadvantage that can be seen is that the ascent with such a low free lift is sluggish, and when moisture accretion is possible, might be insufficient to take the balloon to ceiling.

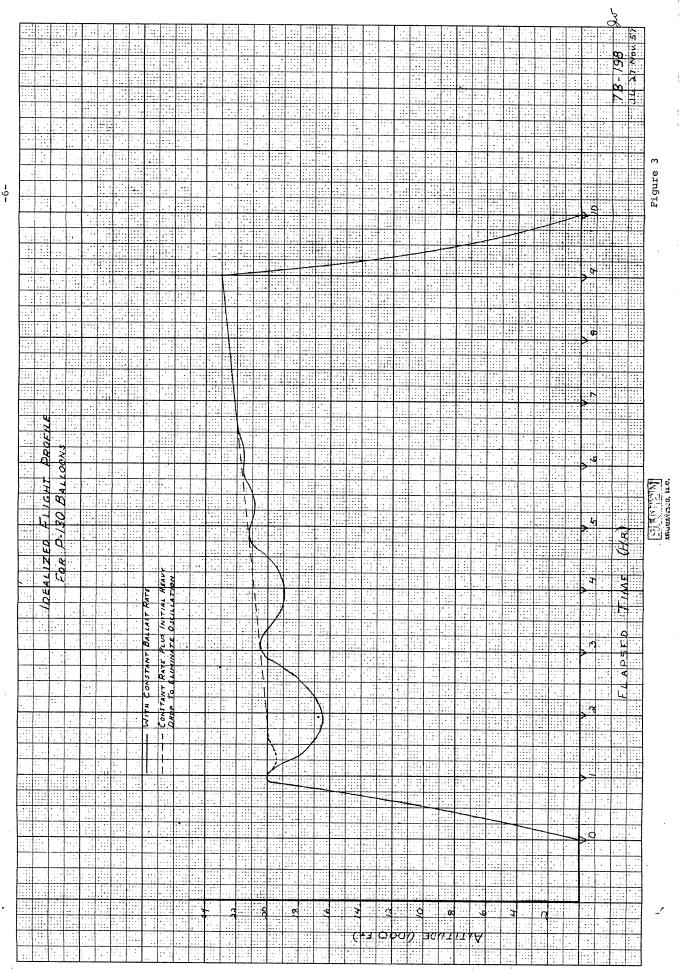
In the second approach, overvalving was counteracted by dropping a relatively large initial quantity of ballast during the period immediately after the balloon reached ceiling. Figure 3 indicates the type of performance to be expected with and without the initial ballast drop. Performance in the latter situation was actually observed in Flight No. 187. Most of the series of flights was devoted to the determination of the amount of initial ballast to be dropped and the delay after launch when this drop should begin. The initial ballast drop denoted on the time-altitude curves refers to weight added to the cards already on the rack (4.76 grams each). This weight was distributed over several cards to allow for variations in rate of rise. In the final flights (P-130's) a weight of 50 grams was distributed on four to six cards with the delay set so that dropping would start just before the attainment of peak altitude was expected. This would allow quick corrective action even if the balloon rose at an unusually high rate. delay times for various free lifts are shown in Figure 4. amount of initial drop required actually will vary with free lift also, however, a standard amount can be adopted with consequent simplification in ballast preparation. The tentative standards are:

For P-130 First 5 cards, 10 grams each (total card wt.) For R-250 First 5 cards, 15 grams each (total card wt.)

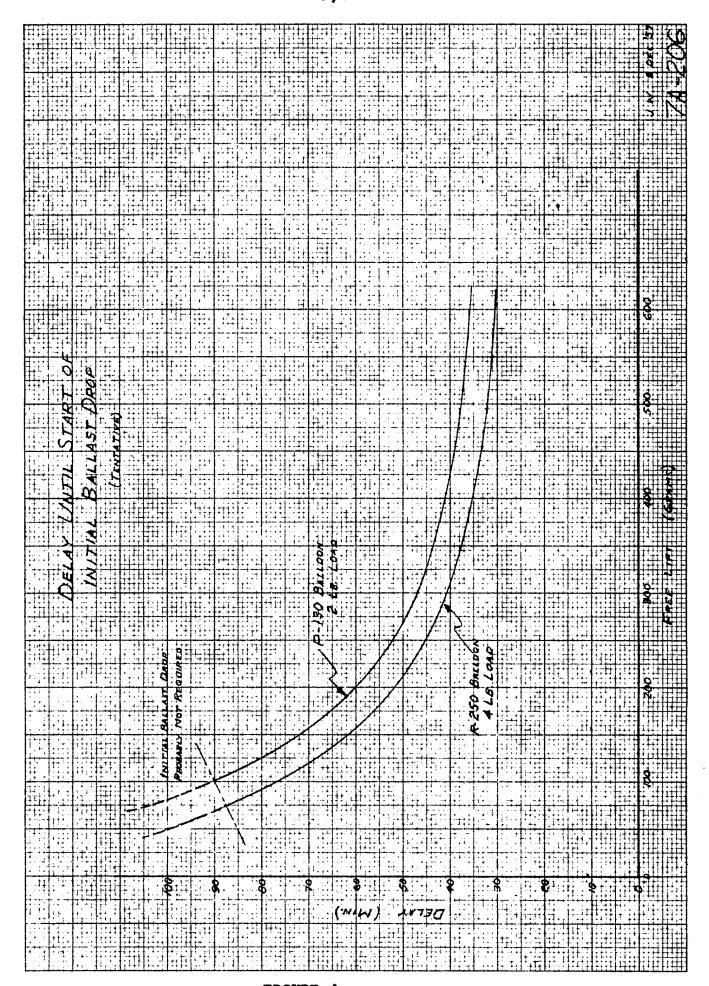
The continuing rate of ballast release after the initial drop is also dependent on some variable, most probably related to cloud cover and variations in radiation influx. As it is virtually impossible to predict the conditions which will be encountered along the flight path, a maximum practical rate should be used on all flights. Rates of 38 and 46 grams per hour were employed in tests. A number of very good flights were obtained with the lower rate, but poor performance was still encountered in some instances with

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the high rate. It is felt that for overall efficiency the ballast cards should not weigh over 6 grams each, which provides a rate of 48 grams per hour. In most cases this should be adequate for both sizes of balloon.

It had been expected that the R-250 would have more natural stability than the P-130. Factors in favor of this hypothesis include:

1. Larger balloon should be more stable.

- 2. Larger appendix should reduce overvalving tendency.
- 3. Floating level is in higher, more stable air mass.

Experience might indicate that the hypothesis is not true, however, and in reconsidering the factors involved, several possible reasons were found.

- 1. A higher rate of rise was experienced than with P-130's, even though the percent of free lift was lower. This would result in a greater overvalving tendency.
- 2. The corrective ballast action taken was the same as for P-130's which proved inadequate. The limited number of R-250 flights did not allow for full exploration of the necessary corrections. In particular, delay times were generally late, so that the post-ascent dip had already begun before ballast release commenced.
- 3. There is a suggestion that the differences in construction between P-130 and R-250 balloons is causing the greaterthan-expected fluctuations in R-250 flights. larger balloon has a considerable amount of slack material in the lower portion where the film is gathered into the base enclosure. It has been noted on large balloons that slack material tends to cause unstable performance because of the ability of said slack material to "belly out", producing a greater, abnormal balloon volume. As the balloon valves gas, it relaxes and reverts to its customary volume which then is too small to support the balloon at its current altitude. This is nothing more than overvalving as has been previously referred to, but would be expected to be greater in this case than for a balloon with a tight skin. It had been assumed that this effect would not be appreciable in the R-250 balloon, however, there is a possibility that it does exist. In any case, with proper corrective action, performance should be at least as good as with P-130 balloons.

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V. RECOMMENDATIONS.

There are several changes which may be made in the timer-ballast unit to improve its reliability and simplify preflight handling.

1. On two flights (Nos. 169, 170) the ballast string was dislodged from its slot resulting in loss of the ballast function. Figure 5 shows the cementing of the ballast string in place as was done on all subsequent flights. A permanent solution to the problem is to pass this string through a diametral hole in the shaft.

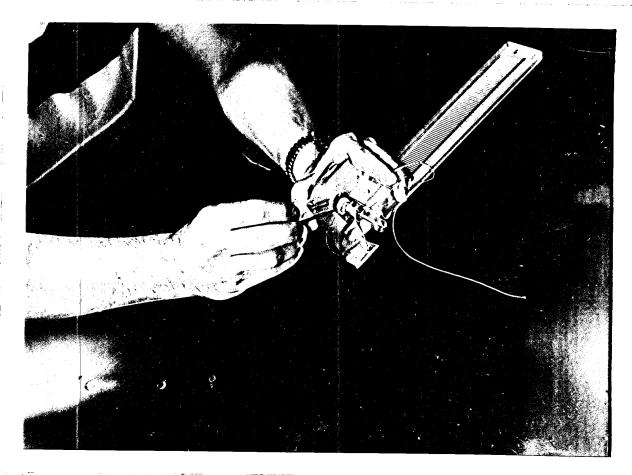


Figure 3

2. The box containing the ballast cards could be constructed so that it would open on a diagonal as shown in Figures 6 and 7. This greatly facilitates loading the cards in the rack.

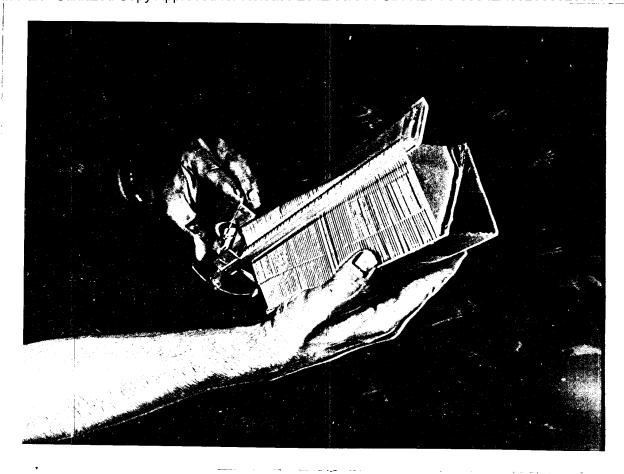


FIGURE 6

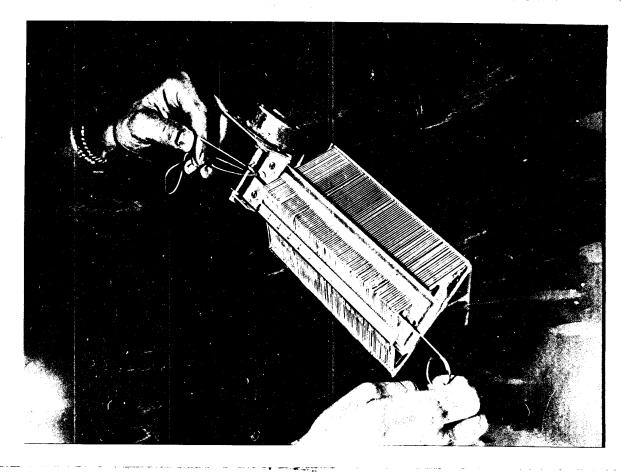


FIGURE 7

-11-

3. A combined snubber line and release link (See Figure 8) has been used to take the place of the load-line-loop which had been affixed to the hub of the timer shaft. This eliminates the possiblity of loose lines fouling the clockworks and provides a convenient method of separating the timer from the balloon. A small inexpensive parachute can be used to slow the descent of the timer.

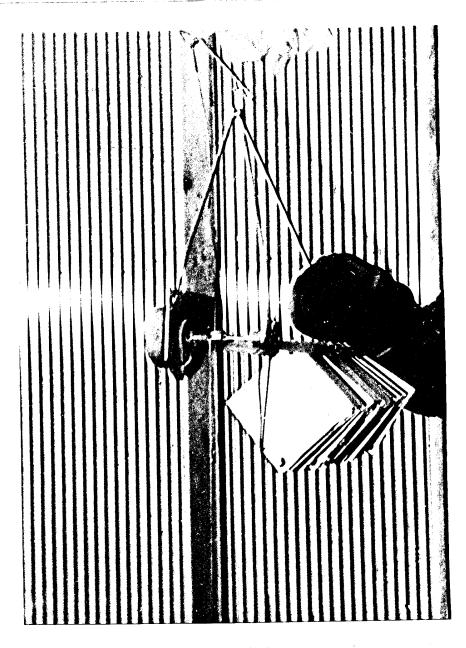


FIGURE 8

4. Setting the timer will probably be simpler if the procedure shown in Figures 9 and 10 is used. Figure 9 shows an indicator mark placed on the ballast string. To set the timer, the shaft is rotated until this mark approaches the desired duration as read on the side scale. Fine adjustment is then made by lining up the shaft-pin (left side of Figure 10) with minute markers placed on the chassis. Small marks could also be made at the end of the ballast rack to indicate the cutting point necessarytto achieve different ballast delays. The unit shown in Figure 10 is set for an 8 hour flight with a 60 minute delay until ballast release starts.

The above procedures permit use of a standard ballast pack for all flights of a given balloon size.

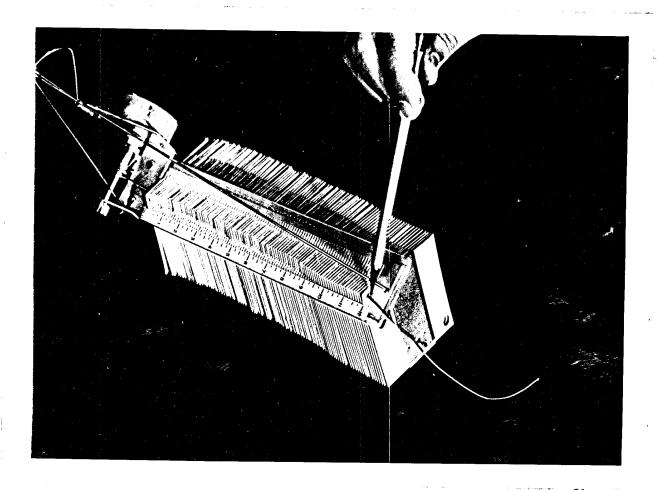


FIGURE 9

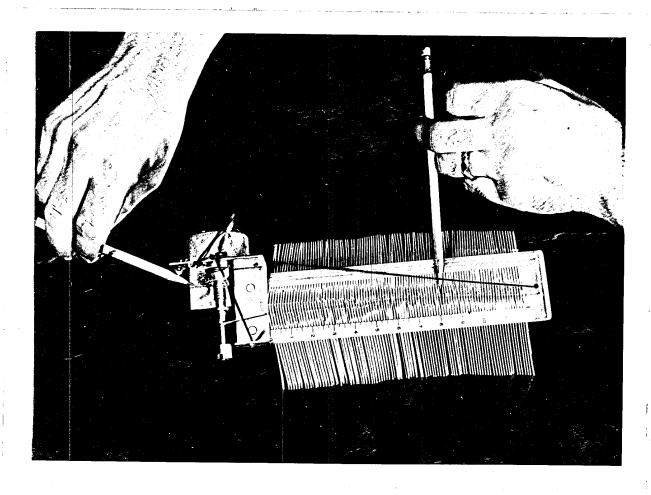


FIGURE 10

- 4. Inflation has been facilitated by the use of a semi-automatic gas control which stops gas flow when inflation is almost complete. Figure 11 shows a balloon at the nearly-complete stage of inflation. A dummy payload, short-coupled to the balloon, is used to reduce the ceiling height necessary for inflation and to make possible the simultaneous preparation of the payload at another location. The gas control shown is not of the ideal dimensions for the particular load train, involved. The dummy weight could include the timer weight if ballast load is standardized.
- 5. Figure 12 shows the balloon immediately prior to launching. The payload has just been attached to the load line with a simple wire link. This picture also shows an intermediate type of release device. The wire pull-pin released the timer from the balloon at flight termination.



FIGURE 11

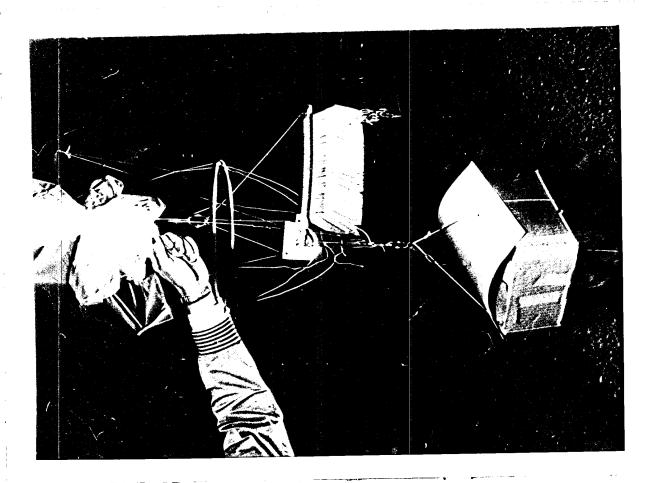


FIGURE 12

VI. SUMMARY.

The analysis of the 41 flights involved in Project 126 and 147, is felt to show conclusively that the timer is satisfactory for the purpose. It has proven to be both accurate and surprisingly dependable. Flights were made through clouds and at night, as well as on clear days. Only one truly unexplained stoppage occurred; all other malfunctions are deemed correctable as outlined in the recommendations.

The ballast action is less certain, but the recommended rates should produce good flights 60-80% of the time and passable performance in the remaining cases. These optimized rates have been tentatively set at:

Balloon	Initital	Steady
	Ballast	Rate
P-130	50 Grams	48 Gm/hr.
R-250	75 Grams	48 Gm/hr.

The rate of ballast release necessary to eliminate all vertical oscillation was not determined, but would probably be prohibitive if any significant payload were to be carried. It is hoped that a simple demand ballast unit may be developed which will provide a high ballast rate when required but will release no ballast when the balloon is floating or rising.

One possible malfunction of the <u>payload</u> release was observed in Flight No. 197. Recovery information plus examination of the returned equipment indicated that the timer had operated properly but the payload did not spill. No proof of where the malfunction occured was available, however.

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	COMMENTS		Ballast string disengaged.	= =	Oscillations late in flight.		expected distance. expect half of filt6 cloud cov. for site. 132, 173, 174		Payload spilled at launch. Timer stopped	Timer stopped early. Reason unknown.	Not recovered. Pt. cloudy Fits. 181-184.	Erratic, moderate oscillations.	Large descent at 5 hr. prob. caused by	clouds. Moderate osc. Timer stopped just before	term. Baro. stopped. Distance indicates proper		and 186. Stopped early. Reason unknown.	Terminated early. Hook malfunction.	Moderate osc. after dip. Timer stopped	as in 184. Moderate osc. damped out.	No ballast dropped. Ballast string not o	Unexplained descent 1 hr. before term.	
NTENANCE	FLOATING	N.G.	N.A.	N.A.	Good 7 hr.	Unk.	Good 3 hr.	Good	N.A.	N.A.	unk.	Fair	Poor	Fair	Unk.	Good	N.A.	" N.A.	Fair	Fair	N.A.	Good 6 hr.	
ALTITUDE MAINTENANCE	POST -ASCENT DIP (Ft.)	12,500	N.A.	N.A.	Neg.	Unk.	None	Neg.	N.A.	N.A.	Unk.	Neg.	1400	1500	3800	2700	N.A.	N.A.	0006	4000	N.A.	1000	
	TIMER PERFORMANCE	OK	: : : : :	Unk.	OK	Prob. OK	OK	OK	Stopped	Stopped	unk.	OK	OK	Stopped	Stopped	Stopped	Stopped	Unk.	Stopped	οĸ	Unk.	OK	
NOI	ACTUAL (HC)	7.95	4.5	3.6	11.9	Unk.	8.0	7.8	5.2	4.0	Unk.	8.0	6.6	1.6#	9.7 #	7.8#	3.5	1.4	7.8#	8.0	3.0	7.0	
DURATION	PLANNED (H.E.)	8	12	12	12	10	ω	œ	æ	α	10	σ,	10	α	10	æ	00	80	60	ω	89	7	_
	DELAY (Min.)	None	None	None	75	75	75	80	75	75	75	75	80	75	75	75	75	75	75	75	75	75	
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BALLAST	INITIAL DROP* (Gm)	None	None	None	None	None	None	30	None	40	None	None	30	27	None	None	None	None	None	None	40	None	
	RATE (@/hr)	38	38	38	38	38	38	38	38	38	38	38	38	38	38	38	38	46	46	46	38	38	
1	FREE LIFT (Gm)	250	200	200	100	100	100	200	200	200	100	100	200	200	200	200	200	700	700	500	250	250	
	BALLOON	R-250	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	P-130	R-250	R-250	
	DATE	29 July	13 Aug	2	22 Aug	3	=	5 Sept	=	± .	12 Sept	=	=	=	24 Sept	:	=	2	:	z	z	=	_
	FLT. NO.	168	169	170	172	173	174	176	177	178	181	182	183	184	186	187	188	189	190	191	192	193	

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		COMMENTS		Baro, not recov. Leaflets at expected	distance. Timer operated, but payload did not spill	Load line loop fouled in timer gears.	Continuing large oscillations.			Descent more severe after initial di	Flts. 203-207 - Night flights.	Not recovered.	Barograph stopped after 1.7 hr.	Osc. damped out. Restarted after su		Large oscillations near end of flight.	Baro, not recov. Leaflets indicate	proper distance. Large osc. near end of flight.	Very erratic. Overcast for Flts. 211-21:	Very erratic.	
MAINTENANCE	FLOATING	PERFORMANCE	Very Good	Unk.	Very Good	N.A.	Poor	N.G.	N.G.	N.G.	Fair	Unk.	Unk.	Fair	Good	Good 5 hr.	Unk.	6000 3 hr.	Poor	Poor	
ALTITUDE		DIP (Ft.)	None	Unk.	Neg.	N.A.	4400	18000	10000	4000	1200	Unk.	7000	8000	None	Neg.	Unk.	None	None	0006	
	TIMER	PERFORMANCE	ØK	Prob. OK	Prob. OK	Stopped	OK	OK	OK	Set wrong?	Set wrong?	Unk.	Prob. OK	Prob. OK	ОК	OK	Prob. OK	OK	ОК	OK	
NOI		ACTUAL (Hr.)	8.6	Unk.	7.8	4.3	7.9	7.5	8.1	0.6	7	Unk.	Unk.	unk.	10	7.8	8.8	Unk.	7.8	7.9	7.9
DURATION		PLANNED (Hr.)	10	10	ω	80	80	ω	00	œ :	თ	α	10	ω	10	80	6	œ	σ.	α	80
		DELAY (Min.)	75	75	75	75	75	75	75	75	09	09	20	20	55	9	09	. 60	9	09	09
ST	DIST-	ION	m	m	т	т	4	4	4	4	4	4	9	4	4	ம	9	ហ	ហ	S	4
BALLAST	INITIM	DROP*	30	30	30	30	40	40	40	40	20	40	57	20	40	20	20	20	20	20	42
	T	E (Hr.)	4 6	46	46	46	38	38	38	38	38	46	38	46	46	46	46	46	46	46	46
	FREE	LIFT (Gm)	5 00	200	200	200	250	250	250	250	250	250	250	250	250	200	200	200	200	200	200
	BALLOON	TYPE	P-130	P-130	P-130	P-130	R-250	R-250	R-250	R-250	R-250	R-250	R-250	R-250	R-250	R-130	R-130	R-130	R-130	R-130	R-130
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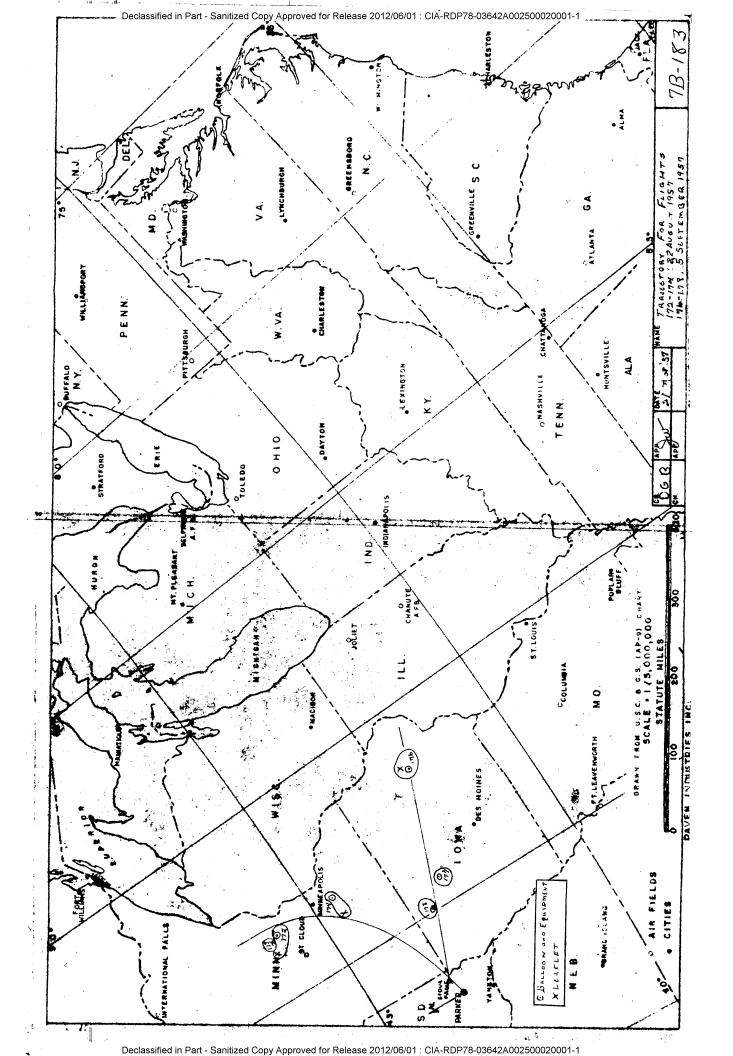
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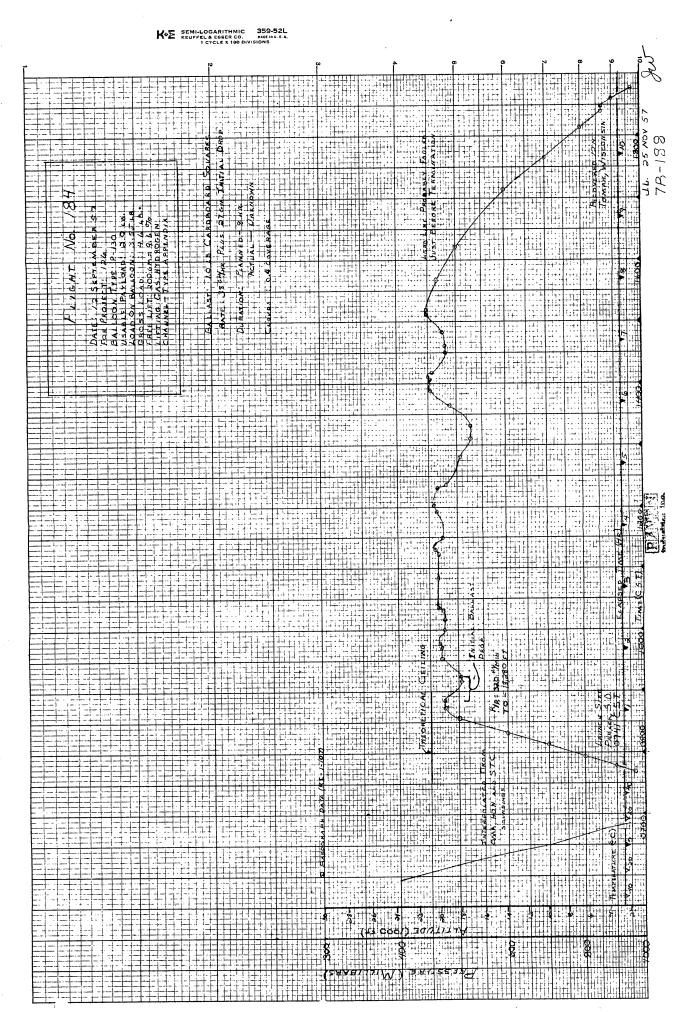
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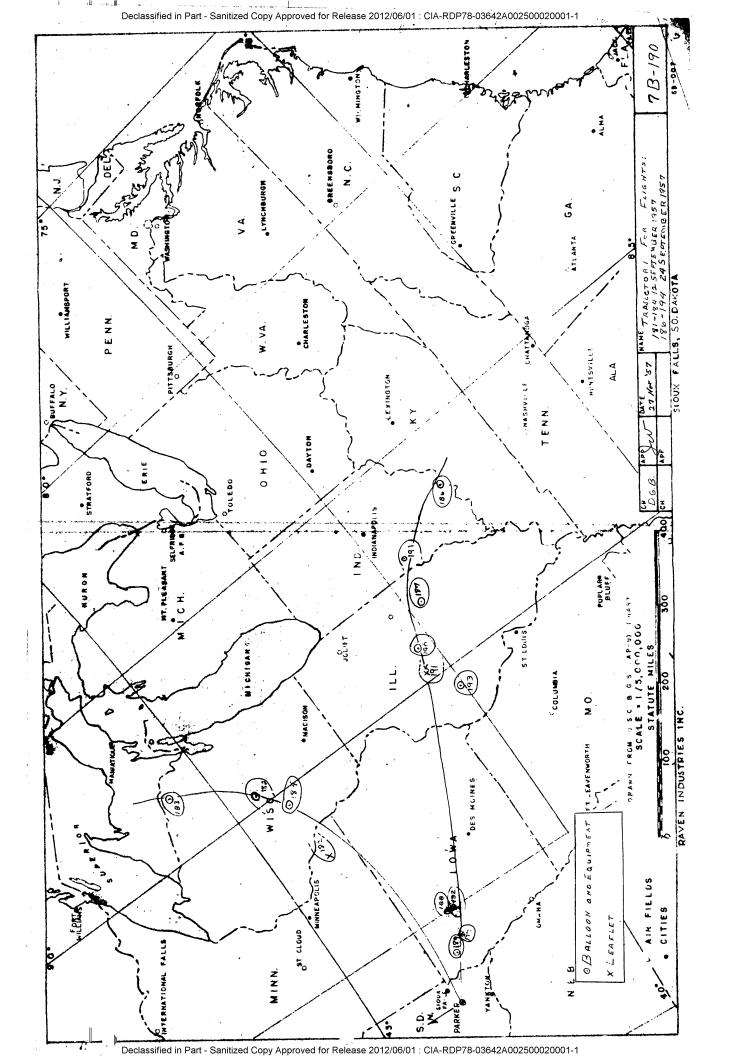
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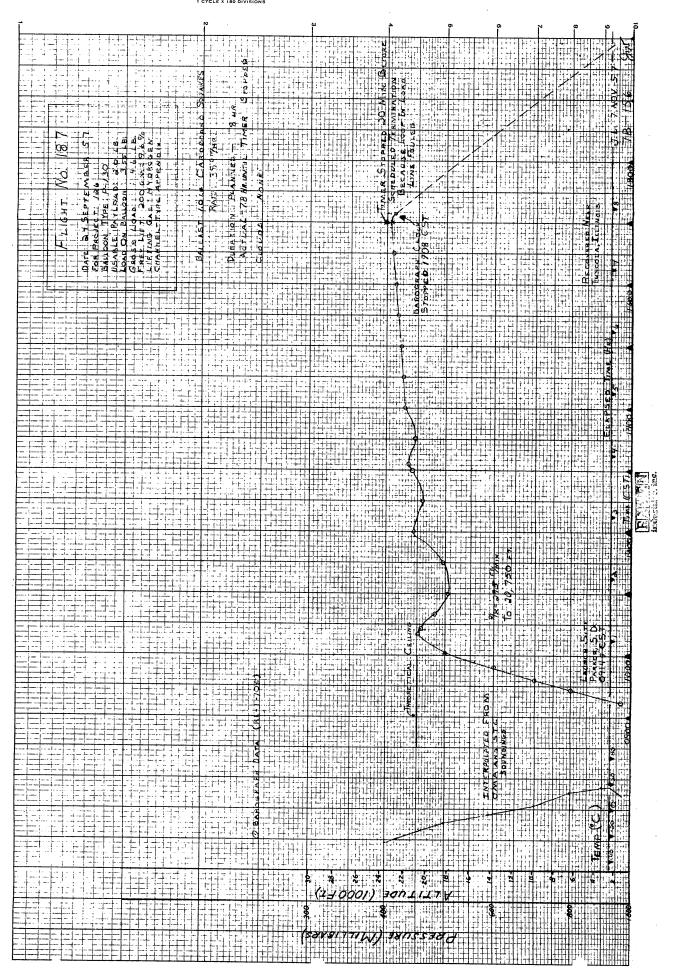


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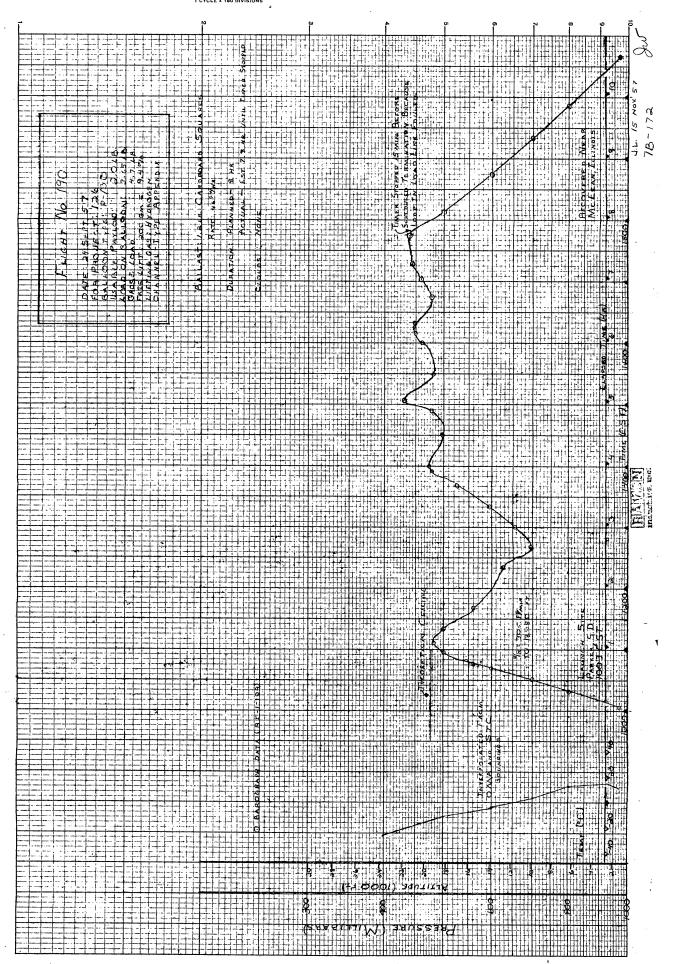
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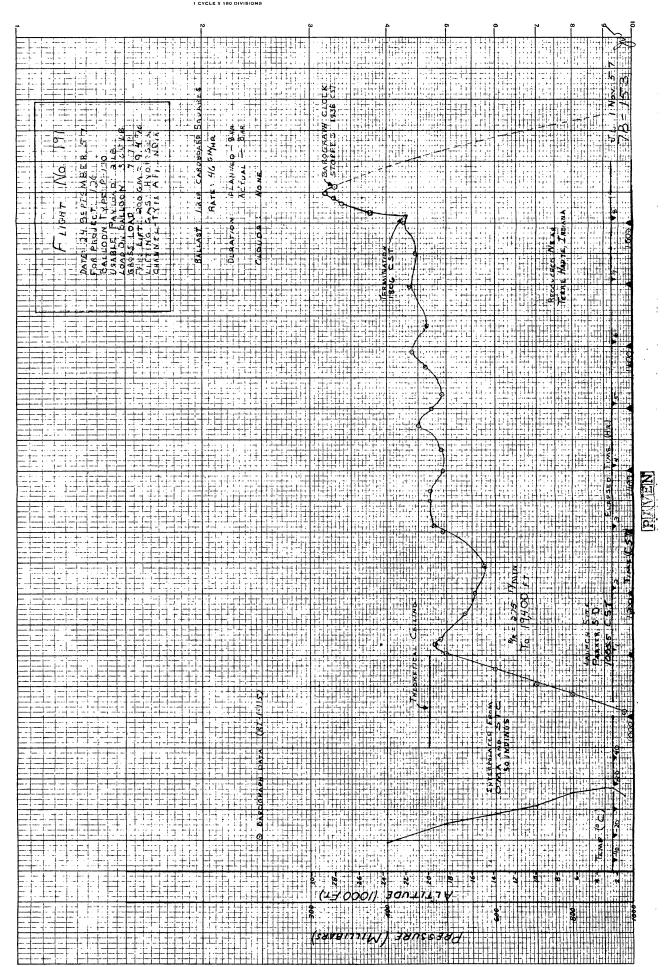
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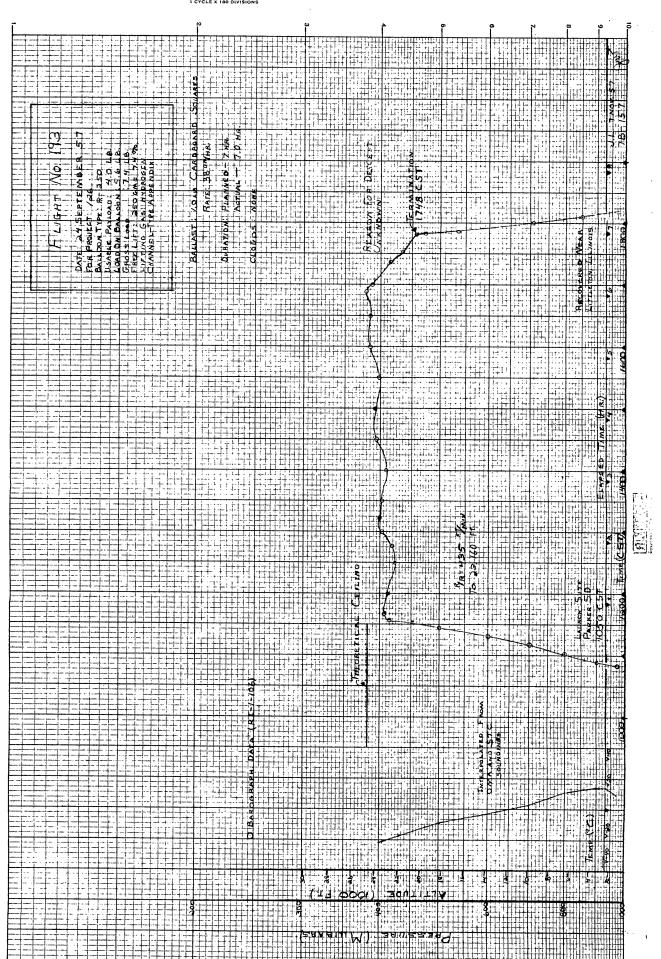
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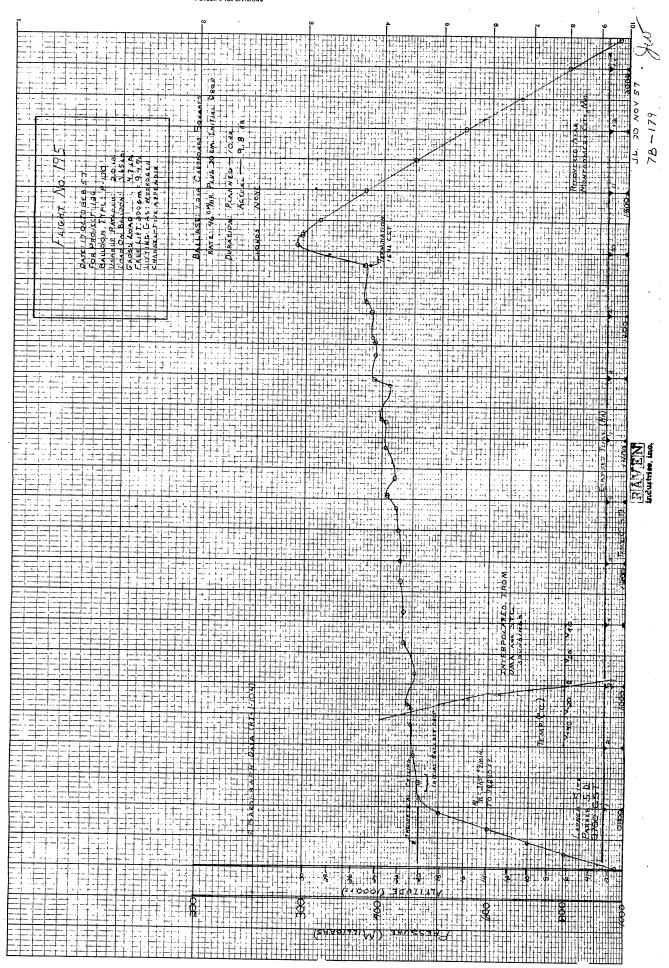
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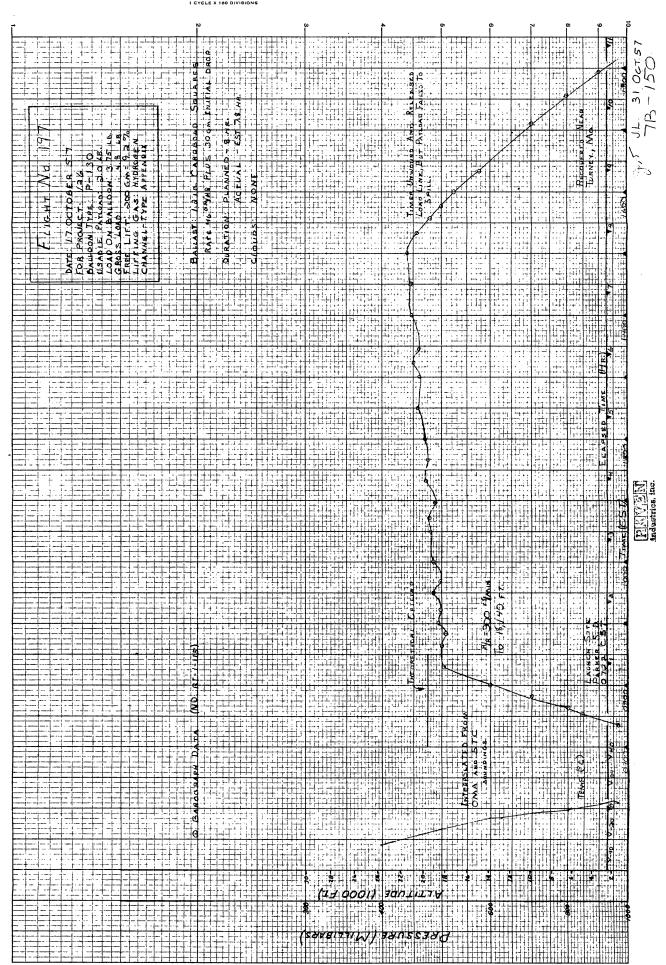
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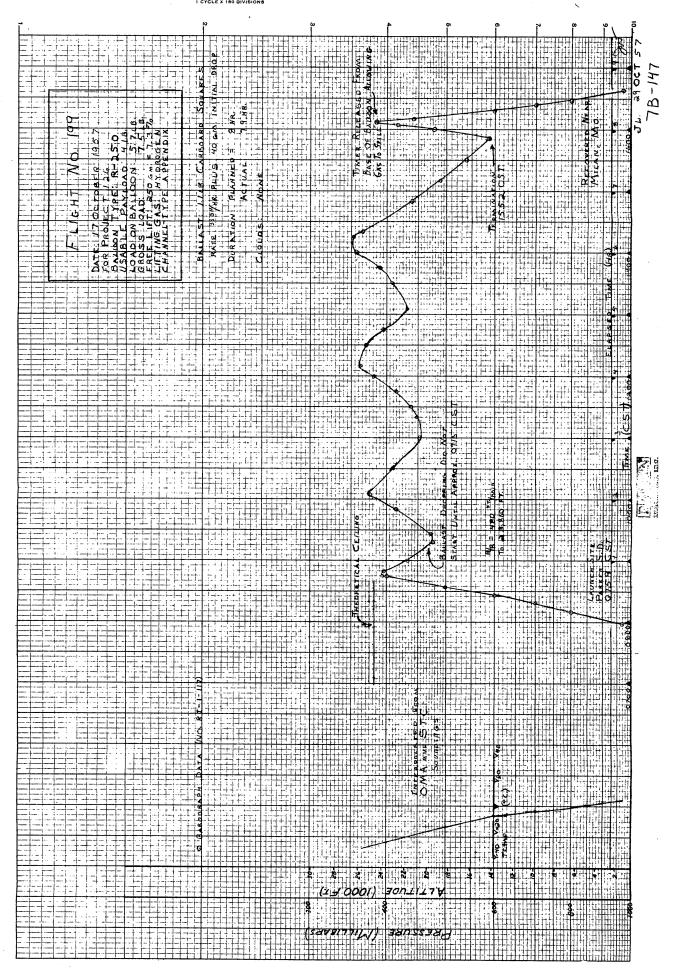
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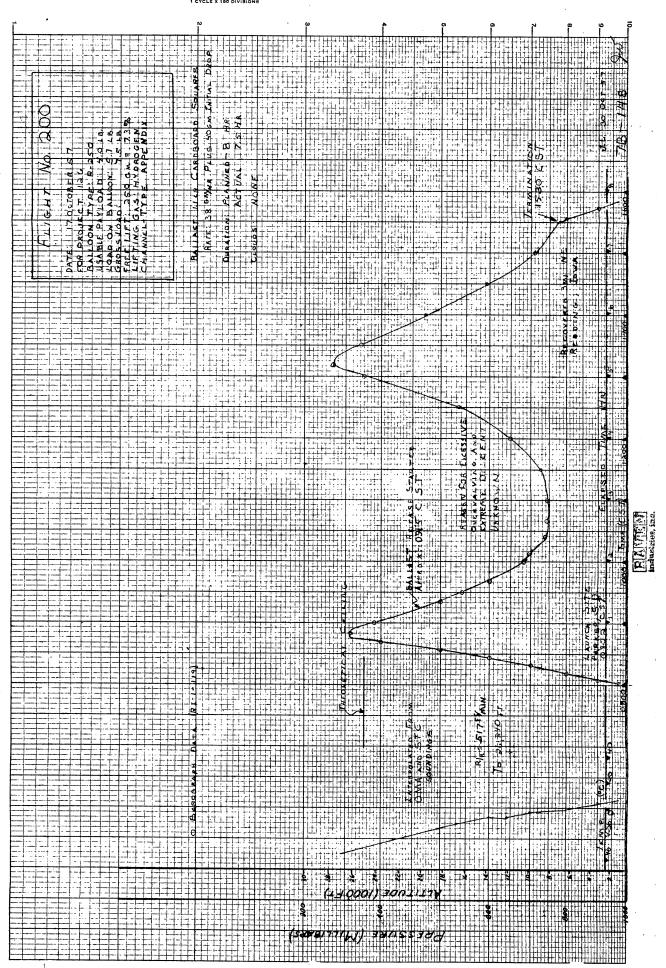
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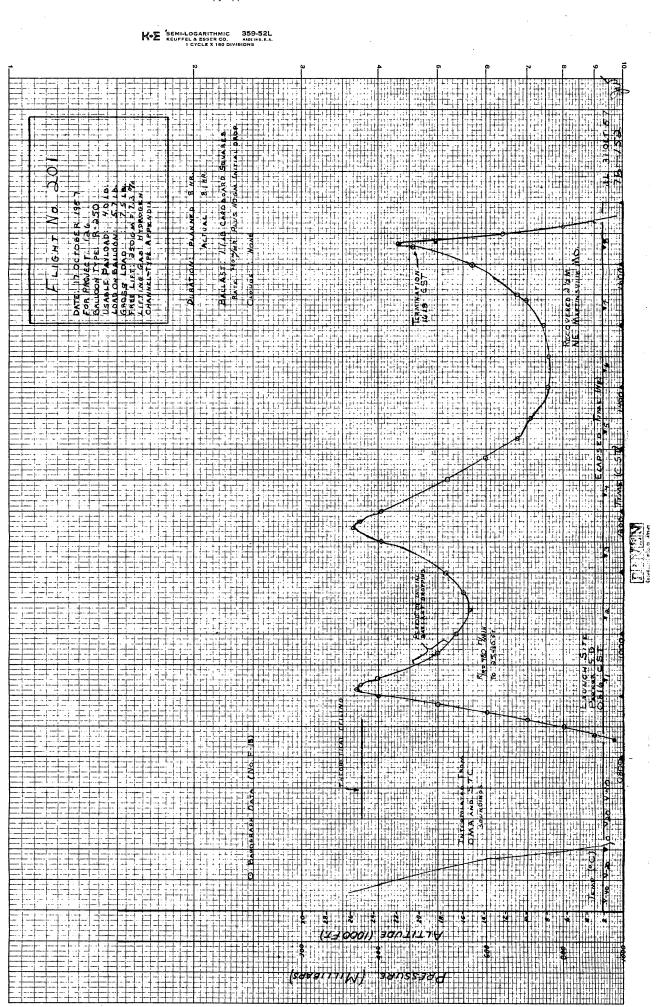
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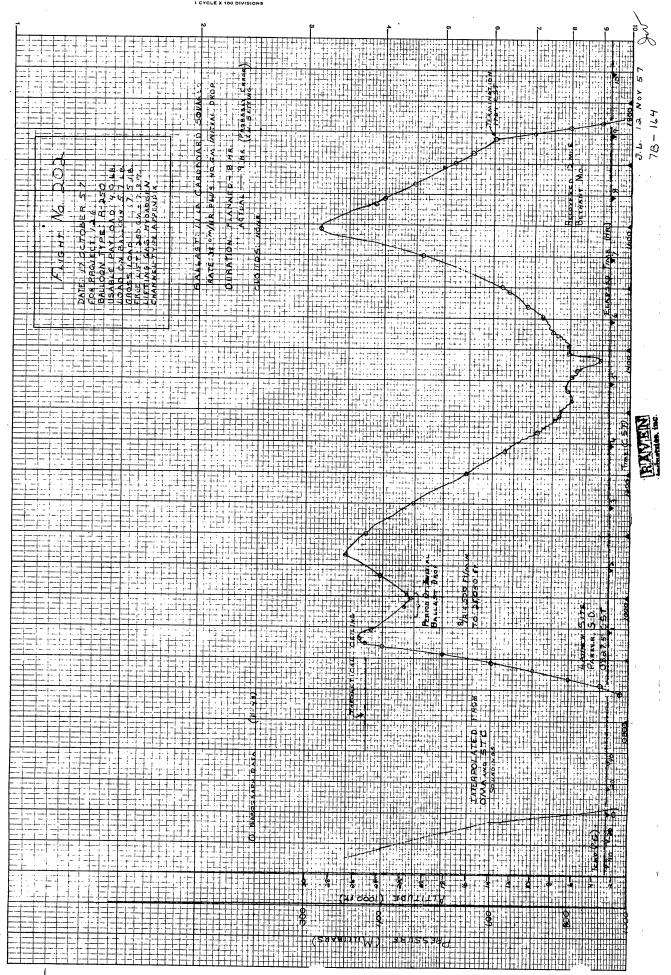
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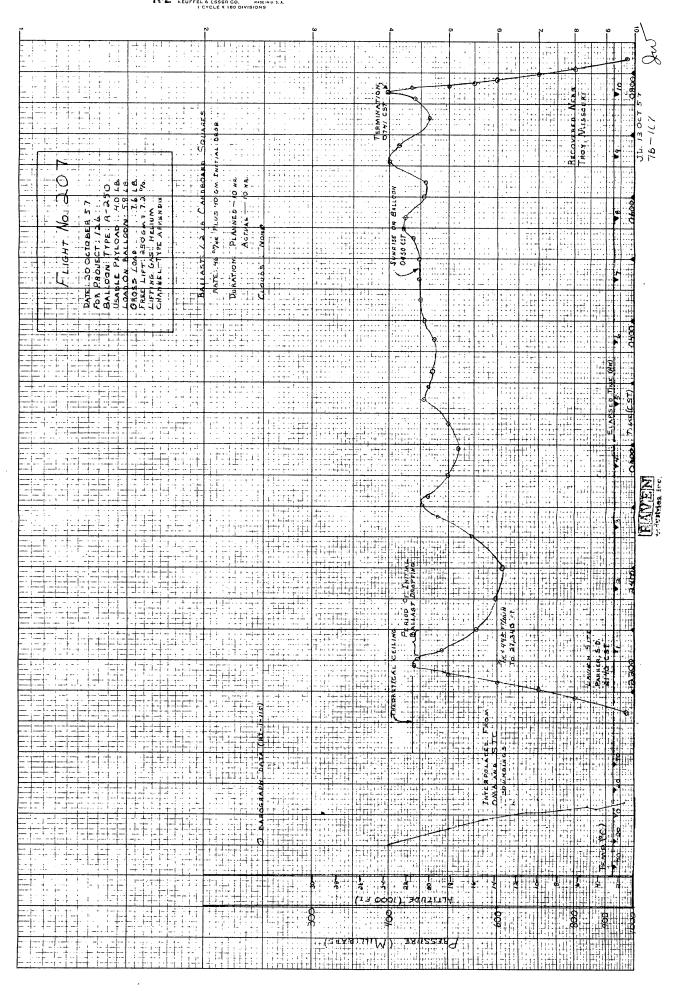
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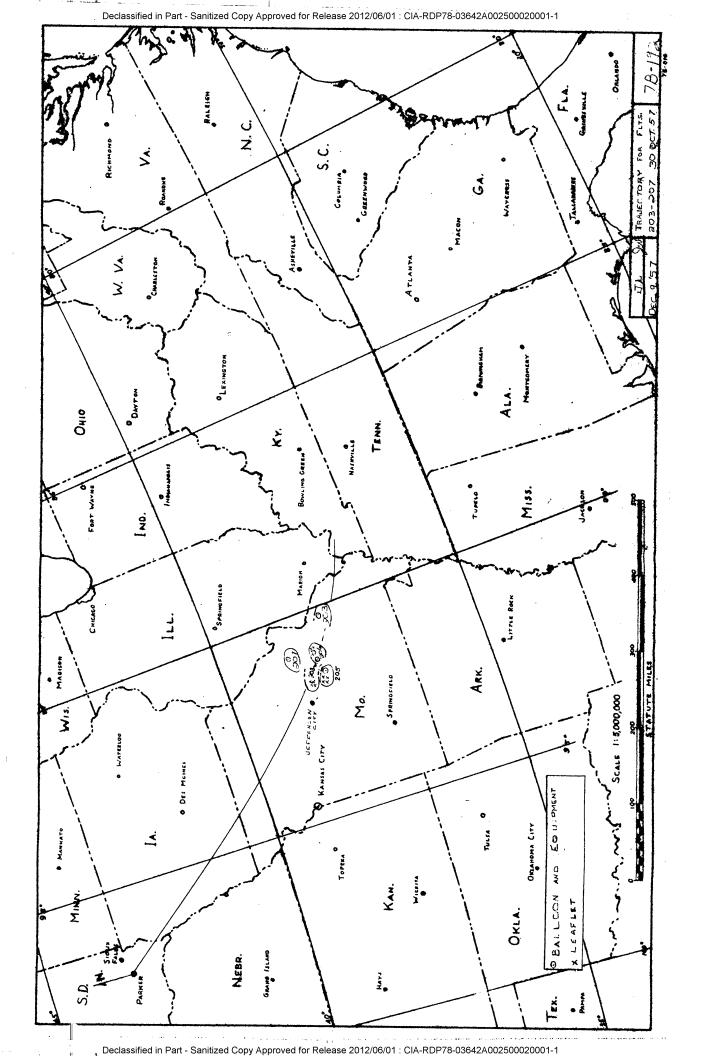
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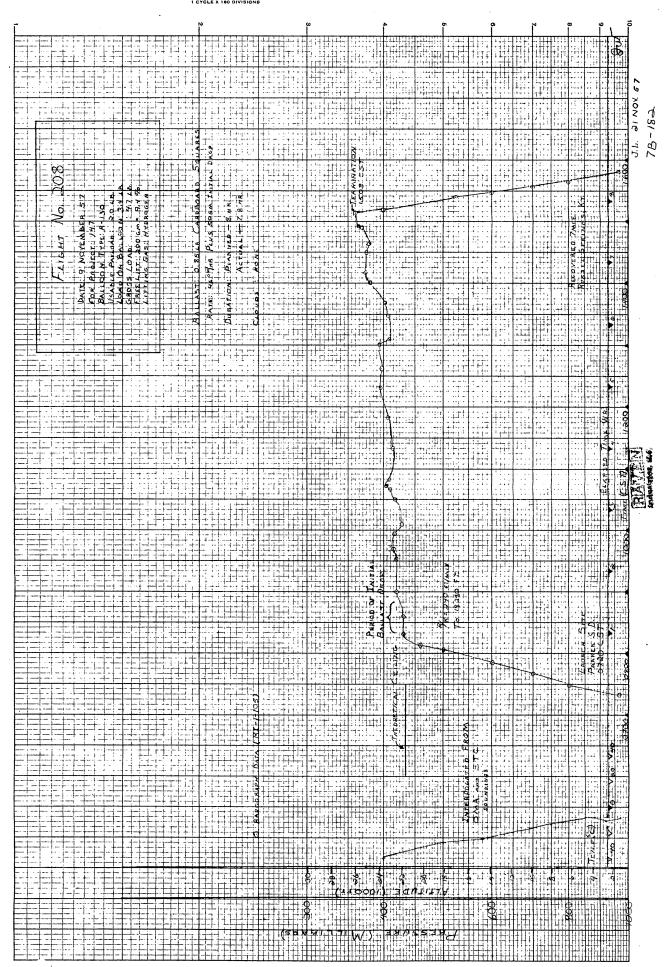
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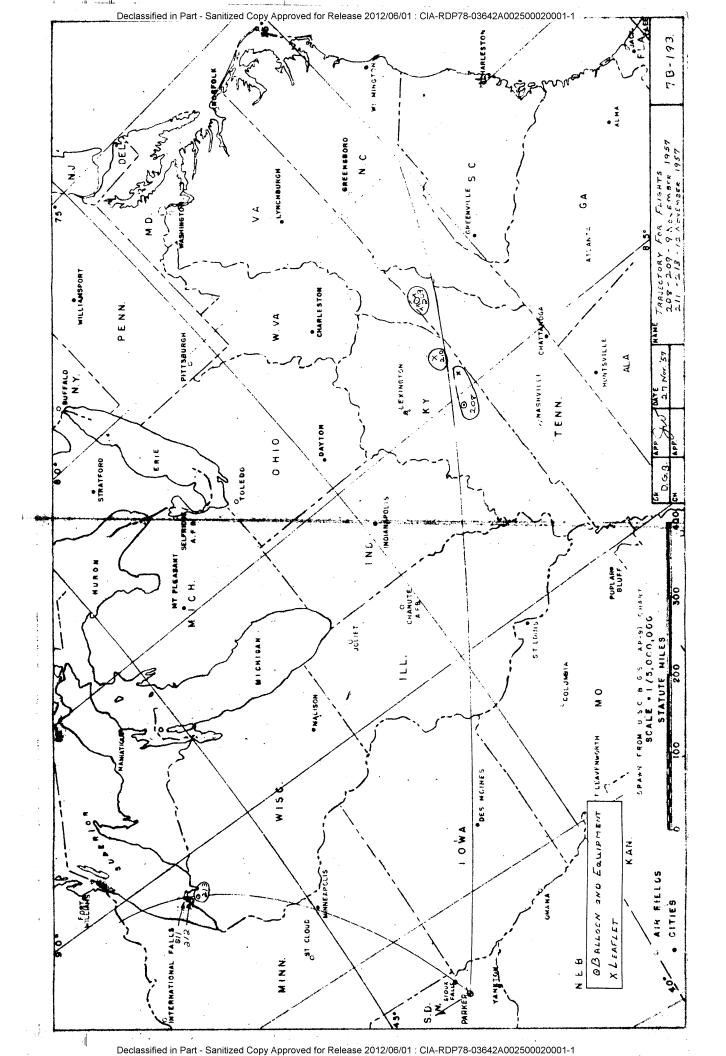


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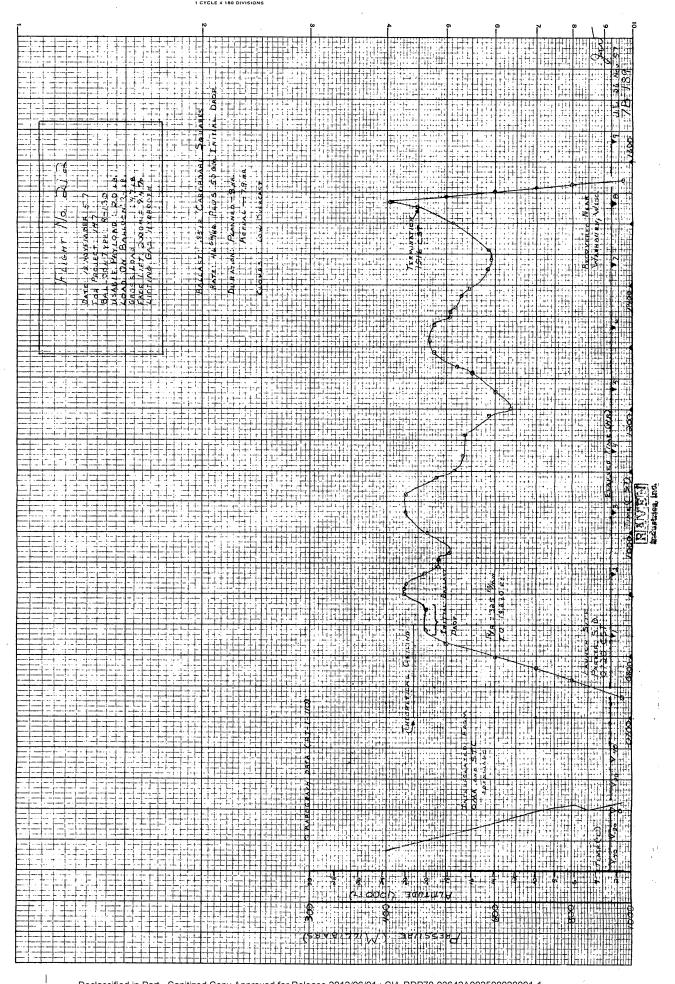
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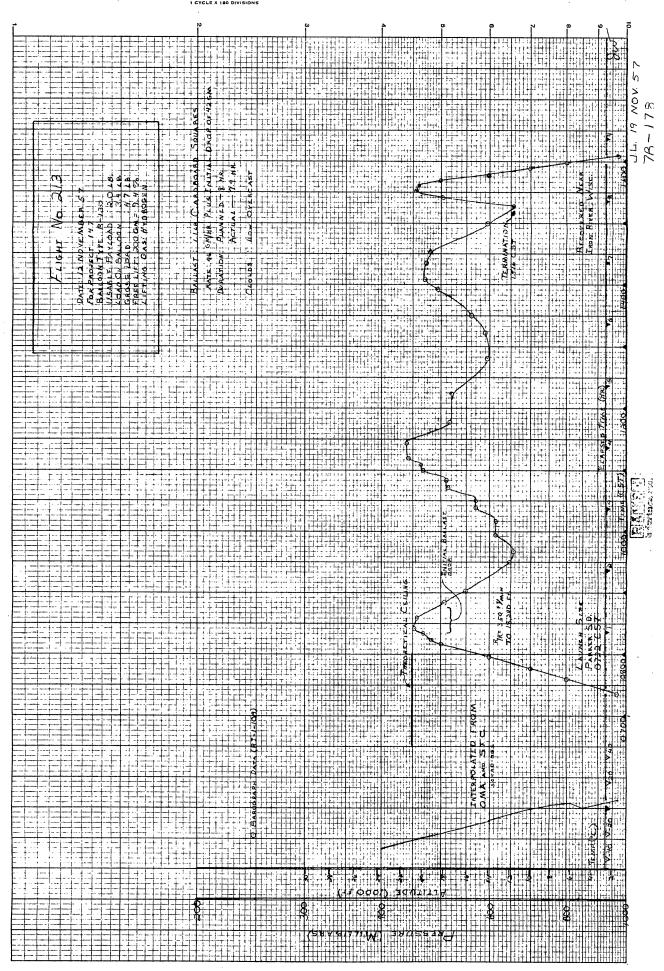
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